

$$r = 1 - \frac{2.312}{99.132} = 0.922$$

$$C_f = \left[ \frac{2 \cdot 1.399 \cdot \left( 0.922^{\frac{1.399-1}{1.399}} - 1 \right)}{(1.399-1) \cdot \left( 0.8^4 - 0.922^{\frac{-2}{1.399}} \right)} \right]^{\frac{1}{2}}$$

$$C_f = 0.472$$

$$C_d = 2.395 \cdot \frac{101325 \cdot \sqrt{1 \cdot 0.0287805 \cdot 8.314472 \cdot 298.15}}{0.472 \cdot 0.01824 \cdot 8.314472 \cdot 99132 \cdot 293.15}$$

$$C_d = 0.985$$

(vi) Calculate the Reynolds number,  $Re^\#$ , for each reference flow rate,  $\dot{Q}_{\text{ref}}$ , using the throat diameter of the venturi,  $d_t$ , and the uncorrected air density,  $\rho$ . Because the dynamic viscosity,  $\mu$ , is needed to compute  $Re^\#$ , you may use your own fluid viscosity model to determine  $\mu$  for your calibration gas (usually air), using good engineering judgment. Alternatively, you may use the Sutherland three-coefficient viscosity model to approximate  $\mu$ , as shown in the following sample calculation for  $Re^\#$ :

$$Re^\# = \frac{4 \cdot \rho \cdot \dot{Q}_{\text{ref}}}{\pi \cdot d_t \cdot \mu}$$

Eq. 1066.625-8

Where, using the Sutherland three-coefficient viscosity model:

$$\mu = \mu_0 \cdot \left( \frac{T_{\text{in}}}{T_0} \right)^{\frac{3}{2}} \cdot \left( \frac{T_0 + S}{T_{\text{in}} + S} \right)$$

Eq. 1066.625-9