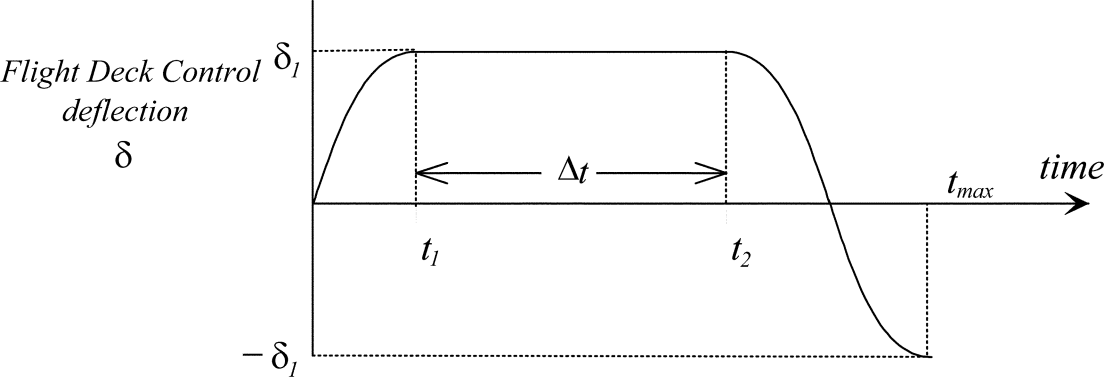
Flight Deck Control deflection
δ
δ1
Δt
time
tmax
t1
t2
− δ1